

Comprehensive Guide to PMDG 737 Engine Start Procedures: From Cold and Dark to Operational Mastery

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The **PMDG 737** series, spanning from the classic NGX to the latest MSFS iterations, represents the pinnacle of high-fidelity flight simulation. For virtual aviators, mastering the engine start procedure of the Boeing 737-800 is a rite of passage that separates casual gamers from serious simulation enthusiasts. This guide provides an exhaustive, technical deep-dive into the pneumatic, electrical, and mechanical processes required to bring the **CFM56-7B** engines to life. Understanding these systems is not merely about following a checklist; it is about comprehending the complex dance of pressure, temperature, and rotational inertia that defines modern jet propulsion.

The Theoretical Framework of Jet Engine Ignition

Before initiating the physical sequence in the cockpit, one must understand the three fundamental requirements for a jet engine start: **Rotational Energy**, **Electricity**, and **Fuel/Ignition**. In the Boeing 737, the engine is not started by an electric motor like a car. Instead, it utilizes a **Pneumatic Starter**. Compressed air is directed into the starter turbine, which is mechanically geared to the N2 (high-pressure compressor) shaft. Once the N2 reaches a specific rotational velocity, fuel and ignition are introduced, leading to self-sustaining combustion.

The Role of the Pneumatic System

The pneumatic system is the backbone of the startup sequence. This system can be pressurized by three primary sources:

- APU (Auxiliary Power Unit) Bleed Air:** The most common source for independent operations.
- Ground Air Cart (Huffer Cart):** Used when the APU is inoperative (MEL item).
- Crossbleed Start:** Using air from an already operating engine to start the second engine.

For a standard start, the **APU Bleed Valve** must be open, and the **Isolation Valve** must be in the 'Auto' or 'Open' position to allow air to reach the engine starter valves on both the left and right sides of the pneumatic manifold. Crucially, the **Air Conditioning Packs** must be turned **OFF**. The packs consume a massive volume of air; leaving them on starves the starter of the pressure needed to rotate the N2 shaft to the required ignition speed.

Technical Specifications and Operational Limits

Monitoring the start requires a keen eye on the Engine Display (ED). Pilots must watch for specific milestones to ensure the engine is not entering a "non-normal" state. The following table outlines the critical parameters during a standard 737-800 engine start sequence.

Parameter	Target/Threshold	Significance
N2 Rotation	Minimum 25% (or max motoring)	Required speed before introducing fuel to prevent a 'Hung Start'.
EGT (Exhaust Gas Temp)	Max 725°C (Start Limit)	Exceeding this indicates a 'Hot Start,' requiring immediate fuel cutoff.
Oil Pressure	Indication by Starter Cutout	Ensures the engine core is lubricated during acceleration.
N1 Rotation	Positive Indication	Confirms the low-pressure spool (the big fan) is rotating freely.
Starter Cutout	~56% N2	The point where the starter disengages and the engine is self-sustaining.

The Electrical Requirement

While the starter is pneumatic, the **Engine Start Valves** and **Ignition System** are electrically controlled. The 737 uses two igniters per engine (Ignition A and B). Typically, pilots alternate between these for each flight to ensure even wear. The **Standby Powersystem** must be operational to ensure that even in a total electrical failure, the igniters can receive power from the battery through an inverter.

Step-by-Step Procedure: Cold and Dark to Engine Idle

This procedure follows the **PMDG 737** standard operating manual, optimized for the Boeing 737-800 variant. It assumes the aircraft is at the gate with the **Pre-Flight Checklist** completed and the APU running.

1. Configuring the Overhead Panel

Before the first blade turns, the pneumatic manifold must be isolated for the start. Navigate to the overhead panel and verify the following:

1. Fuel Pumps: Turn on at least one pump for the tank containing fuel (usually all Aft and Forward pumps for the wing tanks). If the center tank has more than 1000 lbs, turn on those pumps as well.
2. Galley Power: Verify it is ON (standard flow).
3. Air Conditioning Packs: Set both to OFF. This is critical to maximize duct pressure.
4. Isolation Valve: Set to AUTO or OPEN.
5. APU Bleed: Set to ON. Monitor the Duct Pressure gauge; it should read at least 30 PSI for a crisp start.

2. The Ignition Sequence (Engine No. 2 First)

By convention, Engine No. 2 (Right) is started first to check the hydraulic system performance on that side and because it is further from the boarding door.

- Engine Start Switch: Turn to the GRD (Ground) position. This opens the start valve and uses pneumatic pressure to begin rotating the N2 compressor.
- Monitor N2: Watch the Engine Display. You will see N2 increasing. At approximately 20% N2, check for N1 rotation.
- Introduce Fuel: When N2 reaches 25% (or at its maximum motoring speed if 25% isn't reached within a reasonable time), move the Engine Start Lever on the pedestal to the IDLE position.

- Monitor EGT and Fuel Flow: You should see an immediate rise in Fuel Flow followed by a rise in EGT. This indicates Light-off.

3. Starter Cutout and Stabilization

As the engine accelerates, the **Engine Start Switch** will automatically spring back from **GRD** to **OFF** at approximately 56% N2. This is the 'Starter Cutout'. If it fails to snap back, the pilot must manually move it to prevent damage to the starter motor. Once the engine stabilizes, you should see roughly the following 'stabilized' values:

- N1: ~20%
- EGT: ~400°C
- N2: ~60%
- Fuel Flow: ~0.30 kg/h

Repeat the same process for Engine No. 1.

Advanced Comparison: Start Methods

In simulation environments like **PMDG 737 MSFS**, users often encounter different operational scenarios. The following table compares the three main methods of starting the engines.

Feature	APU Start (Standard)	Crossbleed Start	Ground Air Start
Primary Air Source	Internal APU	Operating Engine (>30 PSI)	External Pneumatic Cart
Usage Scenario	Daily Operations	APU failure after 1st start	APU Inoperative at Gate
Complexity	Low	Moderate (Requires thrust mgmt)	Moderate (Requires ground crew)
Safety Risks	Minimal	High (Tailpipe fire risk if low air)	External communication errors

Crossbleed Start Mechanics

If the APU is deferred (not working), pilots must perform a **Crossbleed Start**. After starting Engine 1 via a ground air cart, the cart is disconnected. To start Engine 2, the pilot must increase the thrust on Engine 1 until the duct pressure reaches 30 PSI. The **Isolation Valve** must be **OPEN**. This requires a clear area behind the aircraft as the thrust produced can be significant, potentially causing damage to ground equipment.

Troubleshooting Common Startup Failures

High-fidelity simulations like **PMDG** model 'Non-Normal' events. Understanding how to react to a failed start is a core skill for virtual pilots.

Hot Start

A **Hot Start** occurs when the EGT exceeds the starting limit (725°C). This is usually caused by insufficient airflow to cool the combustion or premature fuel introduction. **Solution:** Immediately move the **Engine Start Lever** to **CUTOFF**. Continue to motor the engine (leave the Start Switch in GRD) to blow out the excess heat and unburnt fuel.

Hung Start

A **Hung Start** is characterized by a failure of the N2 to reach idle speed after fuel is introduced. The engine 'hangs' at a sub-idle RPM. **Solution:** Abort the start by cutting fuel. This is often caused by a low-pressure air source or a faulty starter valve.

No Ignition

If the EGT does not rise within 10 to 15 seconds after moving the fuel lever to idle, you have a **No Ignition** scenario. **Solution:** Check the **Ignition Select Switch**. Switch to the other igniter (e.g., from L to R) and attempt the start again after a cooling period.

The Role of TCAS and Systems Integration

While the focus is on the engines, the integration of other systems is vital. The **TCAS (Traffic Collision Avoidance System)** on the PMDG 737 is typically kept in 'Standby' during the start. However, the electrical transients during the start sequence (switching from APU Gen to Engine Gen) can sometimes cause TCAS or other sensitive avionics to reset if the **Bus Transfer** system is not functioning correctly. Always ensure the **Source Offlights** extinguish after connecting the engine generators to the bus, ensuring a clean electrical handover.

Operational Efficiency and Post-Start Flows

Once both engines are stabilized, the flight crew performs the **After Start Flow**. This involves restoring the aircraft to a 'taxi-ready' configuration:

1. Generators: Turn Engine Gen 1 and 2 to ON (this disconnects the APU from the electrical busses).
2. Probe Heat: Turn ON.
3. Anti-Ice: Set to AUTO or ON if icing conditions exist.
4. Packs: Turn both to AUTO (restoring climate control).
5. APU Bleed: Turn OFF.
6. APU: Stop (if not required for takeoff performance/safety).
7. Engine Start Switches: Set to CONT (Continuous ignition) for takeoff.

Mathematical Considerations: Fuel Flow and EGT

The relationship between Fuel Flow (\$FF\$) and \$EGT\$ during start is roughly linear until stabilization. If the rate of change $\frac{d(EGT)}{dt}$ exceeds a specific internal threshold relative to $\frac{d(N_2)}{dt}$, the PMDG's logic may trigger a warning. Pilots use these mental models to anticipate the 'peak' EGT before it happens. In cold weather, \$N_2\$ will rise slower due to oil viscosity, requiring a longer 'motoring' period before introducing fuel.

Strategic Conclusion

Mastering the engine start of the **PMDG 737** is an exercise in systems engineering. It requires the pilot to be in sync with the pneumatic manifold, the electrical distribution, and the thermodynamic limits of the CFM56 engines. By adhering to the **25% N2** rule, monitoring the **Isolation Valve**, and ensuring the **Packs** are off, virtual pilots replicate the precise professional standards of real-world Boeing crews. This procedural discipline not only ensures the longevity of the simulated engines but also provides the foundational stability required for the safe conduct of the flight. As you move from the gate to the taxiway, the successful roar of two stabilized turbofans serves as the definitive confirmation of a cockpit managed with technical excellence.

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